

Analysis Report

The equipment under test (EUT) is a portable transmitter for a REMOTE CONTROL AIRPLANE operating at 27.145 MHz which is controlled by a crystal. The EUT is powered by six 1.5V AA size batteries. For more detail information pls. refer to the user manual.

Antenna Type: telescope antenna with unique antenna connector

Antenna Gain: 0dBi

The nominal conducted output power specified: -12dBm (+/- 3dB)

The nominal radiated output power (e.r.p) specified: -14.15dBm (+/- 3dB)

Modulation Type: Pulse modulation

According to the KDB 447498:

The worst-case peak radiated emission for the EUT is 83.5dB μ V/m at 3m in the frequency 27.145MHz

The EIRP = $[(FS \cdot D)^2 / 30]$ mW = -11.73dBm

The ERP = EIRP – 2.15 = -13.88 dBm

which is within the production variation.

The maximum conducted output power specified is -9dBm = 0.13mW

The source- based time-averaging conducted output power

= 0.13 * Duty Cycle mW = 0.06 mW < 0.1mW

The SAR Exclusion Threshold Level for 27.145MHz when the minimum test separation distance is < 50mm:

= $474 * [1 + \log(100/f(\text{MHz}))]/2$

= 371.2 mW

Since the source-based time-averaging conducted output power is well below the SAR low threshold level, so the EUT is considered to comply with SAR requirement without testing.

Transmitter Duty Cycle Calculation

The duration of one cycle = 85.2 ms

Effective period of the cycle = 2.0 ms x 1 + 0.6ms x 26 + 1.2ms x 16 = 36.8ms

DC = 36.8 ms / 85.2 ms = 0.432 or 43.2%